



10,000 feet, you might choose blocks centered around 2,000, 5,000, 8,000, and 11,000 feet. Which blocks you choose isn't important because your final climb performance information will be depicted as plots that will show your plane's climb performance at any altitude—even between the test blocks.

Determining your airplane's  $V_X$  and  $V_Y$  airspeeds is one of the test's primary chores, and this means you should fly a bunch of different airspeeds that are slower than  $V_X$  and faster than  $V_Y$ , and the more different airspeeds you fly—the more accurate your plots. For safety, your slowest speed should be sufficiently faster than stall speed and provide the controllability needed to maintain a precise airspeed. Your fastest speed should be only as fast as you intend to fly while climbing. When deciding on your test airspeeds, remember to include your intended cruise-climb speed, the speed you'll fly to change from one cruise altitude to another.

#### Sawtooth Test Procedure

Load the airplane (weight and center of gravity) in the way you'll typically fly it. This may be maximum gross weight, half fuel with just you aboard, or whatever loading you find useful. Record this information.

Take off and, when ready to begin the test, set the altimeter to 29.92, so you can use the altimeter to record the pressure altitude during the test, which you'll use with the outside air temperature (OAT) to compute the density altitude. To make them usable anytime the finished climb charts are based on density altitude, which you can easily find. Not using density altitude means your charts are only good on days that match the test day barometric pressure and temperature.

Fly a couple of hundred feet below the bottom of the test altitude block. Use good judgment here; your lowest block should be at least 1,000

feet above reasonably flat and obstruction-free terrain. You don't need to start your tests by flying the sawtooth climbs in your lowest altitude block. For safety reasons it's smarter to start with a higher block. Save the lower altitude tests for later, when you'll feel more comfortable with the test technique.

In level flight a couple of hundred

feet below your altitude block establish your test climb airspeed. For your first test choose an airspeed close to the predicted  $V_Y$  airspeed. Eventually you'll fly a range of airspeeds, but prudent test pilots start in the middle of the flight envelope—the climb speed envelope in this case. After you've stabilized your airplane in this level-flight condi-

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density altitude) is the same as the other test runs. If this is starting to sound complicated, don't worry. Usually you can complete the entire sawtooth series quickly enough to avoid changing OATs.

Stabilize your plane for the next climb test run at a different airspeed, and repeat the sawtooth procedure until you've completed the climb test for every test airspeed.

#### By the Numbers

1. Load your airplane for the climb test weight and balance.

2. Take off and set the altimeter to 29.92.

3. Establish the climb test condition before climbing through the bottom of the test altitude block. Engine, pitch attitude, airspeed, and trim should be stabilized.

4. Record the time it takes to climb through the altitude block.

5. Descend at idle power below the bottom of the test block, recording OAT at the block midpoint.

6. Repeat Steps 3 through 5 for all planned test airspeeds.

7. Reset your altimeter and land. Review your data for reasonableness, reload your airplane, and repeat the sawtooth climb test profile for the next test altitude block.

#### Data Quality

Good results from your sawtooth climb tests require solid preparation, precise flying, and good observations. The more diligent you are with these, the better your results will be. Here are a few guidelines.

**Airspeed control.** Some of these results are sensitive to airspeed variations. Traditional flight test parameters limit airspeed excursions to a maximum of 1 knot. This may sound unrealistic, but any pilot can achieve it with a little practice and a diligent trim effort.

**Smoothness counts.** Keeping your airspeed  $\pm 1$  knot at the expense of large or abrupt flight control deflections will contaminate your data.

Every time you move a flight control surface you change the airplane's drag. The bigger and faster the surface movement, the greater the drag. Small adjustments are expected, and it's okay to make them within the test altitude block, but make them smoothly.

**Turbulence.** To avoid turbulence perform the test early in the morning or just before dusk. It only takes one bump to invalidate your airspeed or control deflection tolerance

in a small airplane. Avoid flying near rapidly changing terrain that may produce thermals or up/down drafts. You shouldn't be anywhere near clouds.

**Wind.** A dead-calm day is best, but not very realistic. Fly your climbs perpendicular to the wind to avoid any shear effects or transient airspeed indications.

**Pitch attitude reference.** As good as some artificial horizons are, their indications are generally too coarse

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## Test Pilot

for the fine airspeed control required. Fly the test on a clear day with a distinct horizon, and use the horizon to maintain the proper pitch attitude for the climb. A grease pencil mark on the windscreen or side window will help you detect and correct tiny pitch changes before they affect airspeed.

**Straight climb.** Perform each test on a constant heading. Turns while timing your climb will affect your climb performance data.

**Knock it off early.** If you find you're almost set up as you enter the test block you'll probably exceed some tolerance during the test. Rather than forcing the test run and collecting questionable data or abandoning the test halfway through the run, reduce power, descend, and set up again. This will save time and minimize weight changes due to fuel burned.

**Altitude block height.** Accurately timing through a 100-foot altitude block is difficult in an airplane that climbs at 2,000 fpm. Conversely, maintaining the flight tolerance through a 1,000-foot block in an airplane that climbs at 300 fpm is also difficult. Choose a block height that makes sense for your airplane. It should be tall enough that a 1-second timing error won't make a substantial difference in the average rate of climb through the block. And the block should be short enough so there isn't an appreciable difference in climb rate from the bottom to the top. It's okay to have taller blocks at lower altitudes, where the climb rate is better, and shorter blocks at the higher altitudes, because you'll be calculating average climb rates for each block.

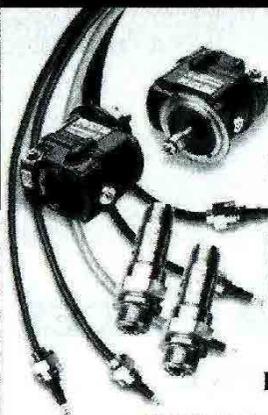
**Take a break.** Relax during your descent between tests; recording the midpoint OAT is your only data obligation. As you descend through the bottom of the test block, begin setting up the next test airspeed. This should expedite establishing the stabilized condition and minimize weight change due to fuel burned.

**Don't trust the VSI.** Feel free to record the reading on the vertical speed indicator within the test block, but use it for correlation with your timed data only. Most VSIs are just too inaccurate for this test.

**Leaning.** Lean your engine as you would for normal climbs. In the airplanes most of us fly, a single mixture setting should accommodate all the runs through the same altitude block. If you must change the mixture between tests in the same block, be as consistent as possible. The same goes for cowl flaps and any other adjustment peculiar to your airplane that can affect climb performance.

**Subjective assessment.** Even if you've flown the test profile within the limits, you may want to make a qualitative comment about the test. You'll know whether you really nailed the point, right on airspeed, pitch attitude set in granite, perfect timing, or whether you pushed the

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limits with airspeed going from 1-knot fast to 1-knot slow four times during the block, constantly searching for the exact pitch attitude, etc. Making such a note on your data card can help explain a wayward data point later. You should also be alert for changes in climb rate within each altitude test block. If climb rate decreases noticeably within the block, the block height is too large. Repeat the test using a smaller, more appropriate altitude block height.

### There's Always Fine Print

You've already done a lot of flying to map your airplane's climb performance, but so far you've tested just one weight, center of gravity, and external configuration. If it will be useful, you can perform these tests for a variety of airplane loadings and configurations. You can test at the maximum and minimum anticipated weights, and interpolate for intermediate weights. Or you may include an intermediate weight in your testing.

In theory the center of gravity location can affect climb performance because it affects the airplane's trim drag. In reality you can ignore this in most homebuilts because its influence is generally minimal.

Open canopies, cowl flaps, and cooling vent scoops affect the airplane's drag. While you can test every conceivable combination, you probably want to spend your flying time differently. Perhaps testing the worst-case combination of weight, center of gravity, and external configuration might be good enough, figuring you'll realize better performance for all other cases. At least this way your planning will be conservative.

Safety is the final and most important point. The tests described require a diligent instrument scan and potentially prolonged nose-high pitch attitudes. Both affect your see-and-avoid capability.

Be careful. Don't perform these tests on a gorgeous Saturday morning near a busy airport. Use your descents between test runs to look around, and why not incorporate a clearing turn into each test setup. Keep an eye on your engine. Low speed climbs at full power tax the engine and inhibit cooling. Remember to fly the airplane first; collect data second.

We described the sawtooth climb test technique this month. Next month we'll use the results from the sawtooth climb tests flown in EAA's Young Eagle's RV-6A to explain data reduction. Then we'll determine the airplane's  $V_Y$ ,  $V_X$ , their respective climb rates and flight path angles, and produce plots that will tell us the RV-6A's climb performance for any climb condition.

Thanks for your comments and suggestions. Keep them coming to Test Pilot, EAA Publications, P.O. Box 3086, Oshkosh, WI 54903-3086 or [editorial@eaa.org](mailto:editorial@eaa.org) with TEST PILOT as the subject of your e-mail.



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